

FMEA  
EMU FAILURE MODE, EFFECT ANALYSISPage: 1  
Date: 03/12/90

01/02/90 SUPERSEDES / /

ANALYST:

NAME P/N QTY	FUNCTION	FAILURE MODE & CAUSES	MISSION PHASE	FAILURE EFFECT	FAILURE DETECTION FLIGHT/GROUND	EFFECT/ ACTIONS	TIME TO CURE	REMARKS/ HAZARD	REF
DISPLAY AND CONTROLS ELECTRONICS, ITEM 350	Provides current limiting for EVC, feedwater solenoid and CTV solenoid power. Provides optical isolation and discrete signal conditioning for DM input discretes and EVC line discretes. Contains battery current and voltage sense circuits, DCN display, and provides secondary power to DCN display, CTV, and sensors.	350PA26: DC/DC converter current limiter fails shorted (Input to output).	PREEVA	END ITEM: Loss of overcurrent protection for DCN for short circuits in DC/DC converter output circuits.	FLIGHT: No.	None required.	3/10	The redundant path is in the SOP. Circuit breakers (current limiters) are standby redundants. Unable to power EMU from either battery or SCU.	
EV702291 (1)	CREW: Electronic component failure.		EVA	Crew INTERFACE: Loss of overcurrent protection for EVC, sensors, and BIDS. A subsequent unprotected short circuit in the ECU would cause the EMU power return P.C. trace in the DCN to fuse, causing a loss of all EMU electrical power.	GROUND: Test. FEMU-R-001, Para. 7.3.5.2.1.1.	TIME AVAILABLE:	A-PASS B-PASS C-PASS		

MISSION:  
None for single  
failure. Terminate  
EVA for subsequent  
failure which  
cause a power  
loss.

CREW/VEHICLE:  
None for single or  
double failure.  
Possible loss of  
crewman with loss  
of SOP.

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ATTACHMENT  
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